## **CLAIMS**

What is claimed is:

- 1. An aircraft fuel system comprising:
- a main fuel pump for pumping fuel to a fuel metering device; and a fuel de-aerator for removing dissolved gases from the fuel before entering said main fuel pump.
- 2. The system of claim 1, wherein removal of said dissolved gases within said fuel reduces a net positive suction pressure required at an inlet of said main fuel pump.
- 3. The system of claim 1, comprising a boost pump for supplying the main fuel pump with fuel at a desired pressure.
- 4. The system of claim 3, wherein said boost pump supplies a net positive suction pressure at an inlet of said main fuel pump.
- 5. The system of claim 3, wherein said fuel de-aerator is disposed between said boost pump and said main fuel pump.
- 6. The system of claim 1, wherein said fuel de-aerator comprises a membrane filter permeable to gases dissolved within said fuel.

- 7. The system of claim 6, wherein said membrane filter is supported on a porous substrate.
- 8. The system of claim 7, comprising a partial pressure differential between a fuel side of said membrane filter and a non-fuel side of said membrane filter, wherein gases diffuse from fuel through said membrane filter to said non-fuel side.
- 9. The system of claim 8, wherein said diffused gases on said non-fuel side are vented overboard.
- 10. The system of claim 7, further comprising a fuel plate defining fuel passages within a housing between an inlet and an outlet
- 11. The system of claim 1, wherein said fuel de-aerator comprises a tubular membrane.
- 12. The system of claim 1, wherein a rate of fuel flow capacity of said system is related to said net positive suction pressure.
- 13. The system of claim 12, wherein said rate of fuel flow capacity increases responsive to removal of gases from said fuel.

- 14. A gas turbine engine assembly comprising:
  - a compressor to compress intake air;
  - a combustor to combust fuel with compressed intake air;
- a turbine section comprising a rotating turbine in flow communication with said combustor; and
- a fuel delivery system comprising a main fuel pump for pumping fuel to a fuel metering device, and a fuel de-aerator for removing dissolved gases from the fuel before entering said main fuel pump.
- 15. The assembly of claim 14, comprising a boost pump for supplying the main fuel pump with fuel at a said net positive suction pressure.
- 16. The assembly of claim 15, wherein said fuel de-aerator is disposed between said boost pump and said main fuel pump.
- 17. The assembly of claim 14, wherein said fuel de-aerator comprises a membrane filter permeable to gases dissolved within said fuel.
- 18. The assembly of claim 17, comprising a partial pressure differential between a fuel side of said membrane filter and a non-fuel side of said membrane filter, wherein gases diffuse from said fuel side through said membrane filter to said non-fuel side.

- 19. A method of improving fuel system operational capacity comprising the steps of:
  - a) flowing fuel containing dissolved gases through a de-aerator;
  - b) removing dissolved gases from fuel within the de-aerator; and
- c) flowing fuel to an inlet of a main fuel pump at a required net positive suction pressure.
- 20. The method of claim 19, further comprising flowing fuel along a fuel side of a permeable membrane within the de-aerator.
- 21. The method of claim 20, comprising providing a partial pressure differential between the fuel side of the permeable membrane and a non-fuel side to diffuse dissolved gases from the fuel through the permeable membrane.
- 22. The method of claim 21, comprising pumping fuel into the de-aerator with a boost pump to provide the required net positive suction pressure to the inlet of the main pump.